PROBLEM SOLUTION:

Normal and shear stresses and their directions

Younus Ahmed Khan

Problem-3 with solution

Consider a "two-dimensional" state of stress in the x-y plane characterized by:

 $\sigma_{xx} = 2,500$ $\sigma_{yy} = 5,200$ $\tau_{xy} = 3,700$

where units are psi and tension is positive. Find: the magnitude and direction of the principal stresses and illustrate with a sketch.

Given: 2D stress state with tension (+) $\sigma_{xx} = 2,500$ $\sigma_{yy} = 5,200$ $\tau_{xy} = 3,700$ and units in psi.





find: the magnitude and direction of the maximum shear stress and illustrate with a sketch.

$$\sigma_{xx} = 2,500 \quad \sigma_{yy} = 5,200 \quad \tau_{xy} = 3,200 \text{ psi}$$

(tension+)
Find: τ_{max} .

Solution:

From notes

$$\tau_{\text{max}} = \left[\left(\frac{\sigma_{xx} - \sigma_{yy}}{2} \right)^2 + (\tau_{xy})^2 \right]^{1/2}$$
$$= \left[\left(\frac{2,500 - 5,200}{2} \right)^2 + (3,200)^2 \right]^{1/2}$$
$$\tau_{\text{max}} = 3,939 \text{ psi}$$

$$\tan 2\alpha^{**} = \frac{-\frac{1}{2}(\sigma_{xx} - \sigma_{yy})}{\tau_{xy}}$$
$$= \frac{-\left(\frac{2,500 - 5,200}{2}\right)}{3,700}$$
$$\tan 2\alpha^{**} = 0.36486$$
$$2\alpha^{**} = 20$$
$$\alpha^{**} = 10^{\circ}$$
or $\alpha^{**} = 100^{\circ}$



Problem-4 with solution

Consider a "two-dimensional" state of stress in the x-y plane characterized by:

 $\sigma_{xx} = 17.24$ $\sigma_{yy} = 35.86$ $\tau_{xy} = 25.52$

where units are MPa and tension is positive. Find: the magnitude and direction of the principal stresses and illustrate with a sketch.





find: the magnitude and direction of the maximum shear stress and illustrate with a sketch.

Given: Problem 13 data: $\sigma_{xx} = 17.24 \text{ MPa}$ $\sigma_{yy} = 35.86 \quad \tau_{xy} = 25.52$ Find: τ_{max}, α^{**} .

Formulas

$$\tau_{\max} = \left[\left(\frac{\sigma_{xx} - \sigma_{yy}}{2} \right)^2 + (\tau_{xy})^2 \right]^{1/2}$$
$$= \left[\left(\frac{17.24 - 35.86}{2} \right)^2 + (25.52)^2 \right]^{1/2}$$
$$\underline{\tau_{\max}} = 27.17 \text{ MPa}$$





$$\tan 2\alpha^{**} = \frac{-\frac{1}{2}(\sigma_{xx} - \sigma_{yy})}{\tau_{xy}}$$
$$= \frac{-\frac{1}{2}(17.24 - 35.86)}{25.52}$$
$$\tan 2\alpha^{**} = 0.3648$$
$$2\alpha^{**} = 20.0^{\circ}, 200^{\circ}$$
$$\alpha^{**} = 10^{\circ}, 100^{\circ}$$

Problem-5 with solution

Suppose that

 $\sigma_{xx} = 2,500$ $\sigma_{yy} = 5,200$ $\tau_{xy} = 3,700$

and the z-direction shear stresses (τ_{zx} , τ_{zy}) are zero, while the z-direction normal stress (σ_{zz}) is 4,000 psi. Find: the major, intermediate, and minor principal stresses.

Given:
$$\sigma_{xx} = 2,500$$
 $\sigma_{yy} = 5,200$ $\tau_{xy} = 3,700$ psi and $\tau_{xz} = 0$ $\tau_{yz} = 0$ $\sigma_{zz} = 400$ psi Find: $\sigma_1, \sigma_2, \sigma_3$.

Solution:

Assume tension is positive. The plane that σ_{zz} acts on is shear-free and therefore is a principal plane. Thus, σ_{zz} is a principal stress. Also (from prob. 11) in the *xy*-plane

$$\sigma_1 = 7,789 \& \sigma_3 = -89 \text{ psi}$$

 $\therefore \sigma_1 = 7,789, \ \sigma_2 = 4,000(\sigma_{zz}) \& \sigma_3 = -89 \text{ psi}$

Find: The maximum shear stress and orientation of associated plane.

Solution:

Since $\tau_{xz} = \tau_{yz} = 0$, it seems $\frac{\tau_{\max} = 3,939 \text{ psi} : \left[= \frac{1}{2}(\sigma_1 - \sigma_3) \right]}{\underline{\alpha^{**} = 10^\circ}} \therefore \text{ must see in } x \text{-} y \text{ view}$



Problem-6 with solution

Suppose that

 $\sigma_{xx} = 17.24$ $\sigma_{yy} = 35.86$ $\tau_{xy} = 25.52$

in MPa and the z-direction shear stresses (τ_{zx} , τ_{zy}) are zero, while the z-direction normal stress (σ_{zz}) is 27.59 MPa. Find: the major, intermediate, and minor principal stresses.

Given: $\sigma_{xx} = 17.24 \text{ MPa}$ $\sigma_{yy} = 35.86 \text{ MPa}$ $\sigma_{zz} = 27.59 \text{ MPa}$ $\tau_{xy} = 25.52$ $\tau_{yz} = 0$ $\tau_{zx} = 0$ Find: $\sigma_1, \sigma_2, \sigma_3$.

Solution:

Assume tension is positive. The plane that σ_{zz} acts on is a shear-free plane $(\tau_{yz} = \tau_{zx} = 0)$ and therefore is a principal plane. Thus, σ_{zz} is a principal stress.

Formulas:
$$\begin{pmatrix} \sigma_1 \\ \sigma_3 \end{pmatrix} = \frac{\sigma_{xx} + \sigma_{yy}}{2} \pm \left[\left(\frac{\sigma_{xx} - \sigma_{yy}}{2} \right)^2 + (\tau_{xy})^2 \right]^{1/2}$$

 $\sigma_1 = 53.72 \text{ MPa}$
 $\sigma_3 = -0.62 \text{ MPa}$ in x-y plane

$$\sigma_2 = \sigma_{zz} = 27.59 \text{ MFa}$$

 $\therefore \sigma_1 = 53.72, \sigma_2 = 27.59, \sigma_3 = -0.62 \text{ MPa}$

Find: Maximum shear stress and orientation

Solution:

$$\tau_{max} = 27.17 \text{ MPa } y$$

Formulas

$$\tau_{\max} = \left[\left(\frac{\sigma_{xx} - \sigma_{yy}}{2} \right)^2 + (\tau_{xy})^2 \right]^{1/2}$$
$$\tan 2\alpha^{**} = \frac{-\frac{1}{2}(\sigma_{xx} - \sigma_{yy})}{\tau_{xy}}$$

Because
$$\tau_{yz} = \tau_{zx} = 0$$

and $\tau_{max} = \frac{1}{2}(\sigma_1 - \sigma_3)$
 $\frac{\tau_{max} = 27.17 \text{ MPa}}{\alpha^{**} = 10^\circ}$